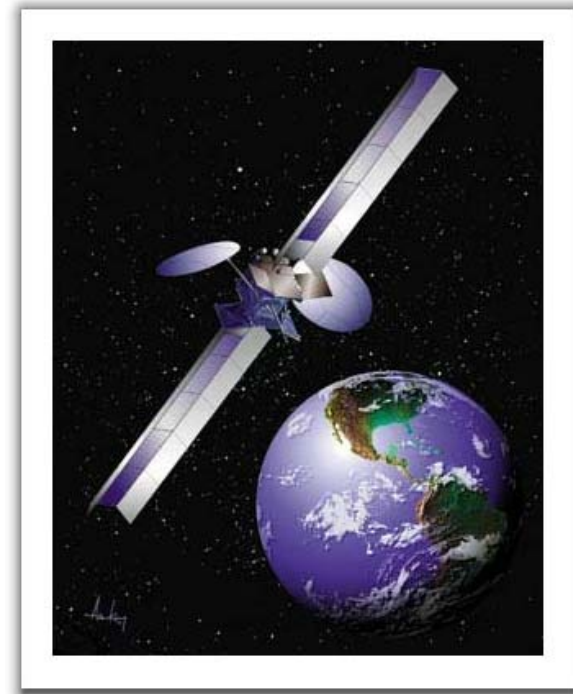


An Introduction to Satellite Communication Systems

Derrick Cheng
September 30, 2003



Some data about satellite location

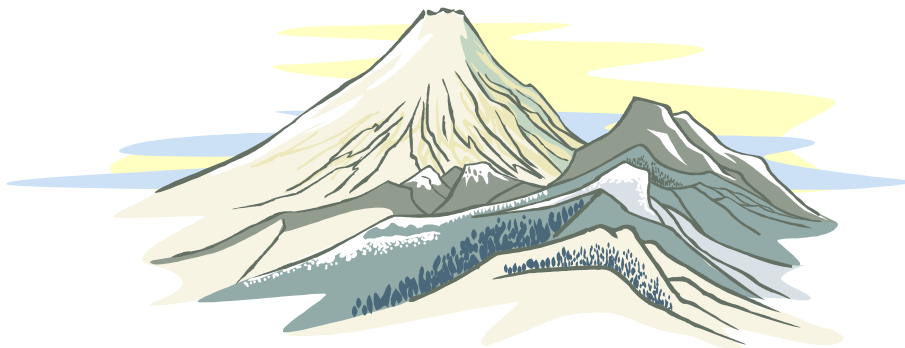
Satellite at Low Earth Orbit: 800 km



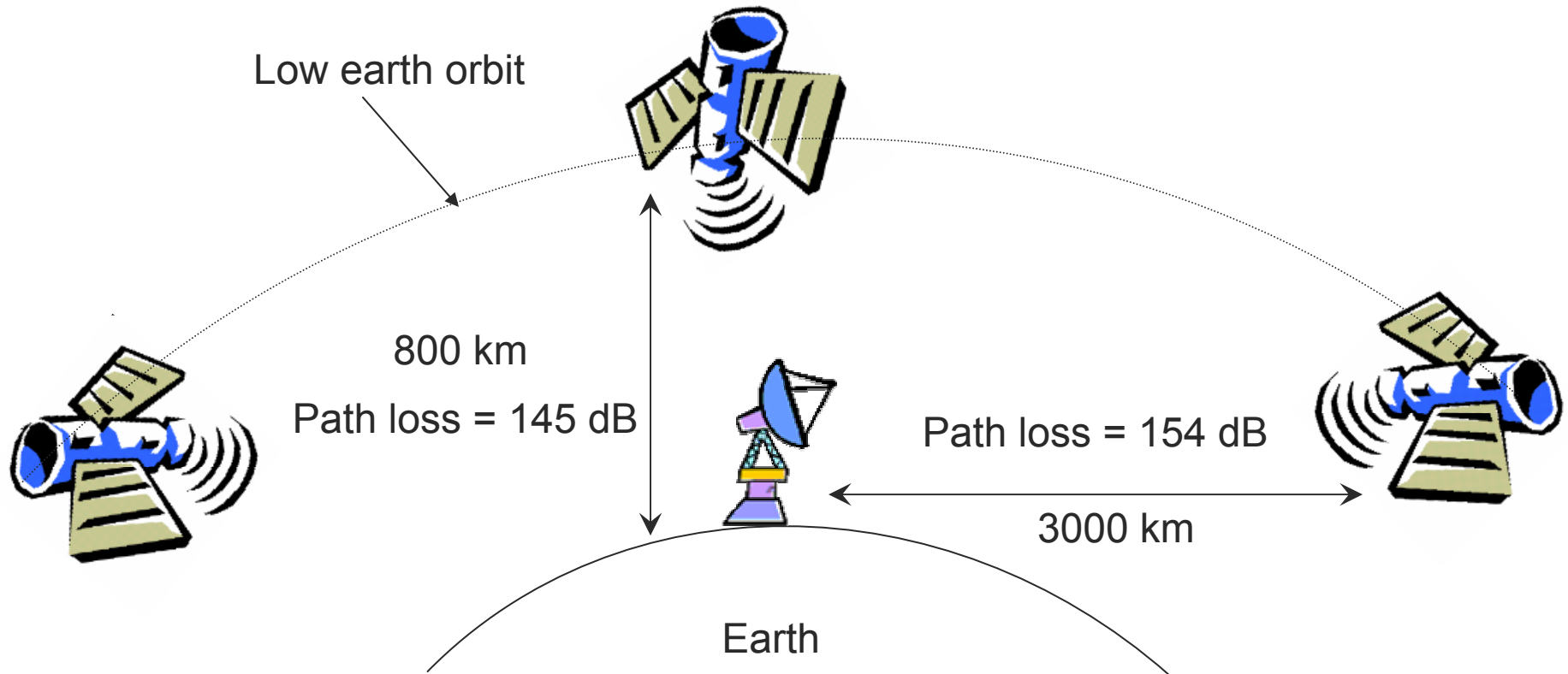
Aircraft: 20 km



Everest: 9 km



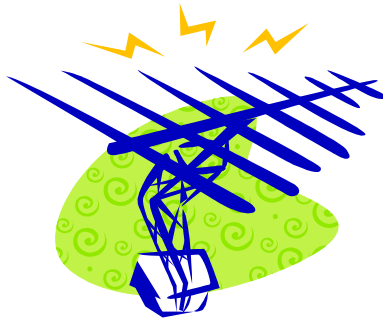
More data about satellite location



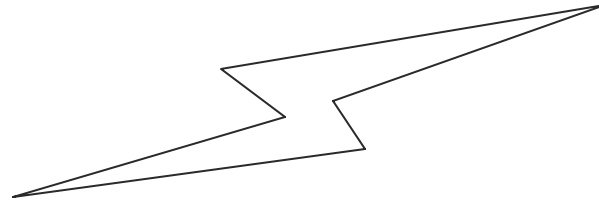
Path loss based on transmission frequency: 435 MHz

How does it work?

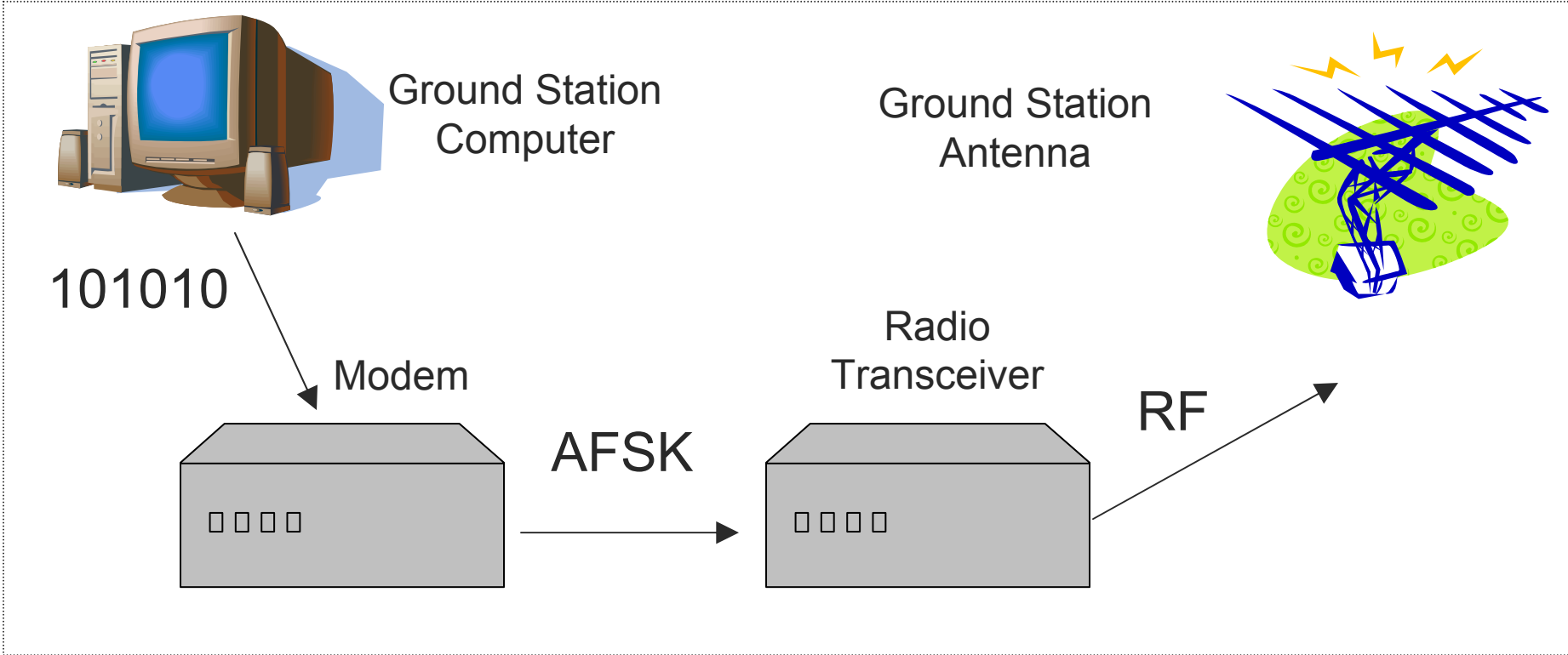
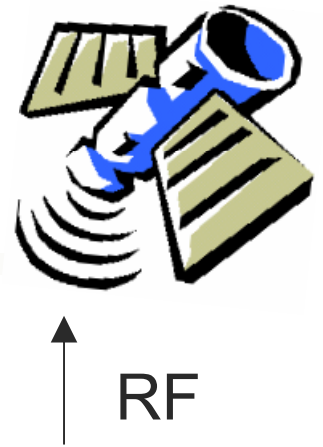
Ground Station

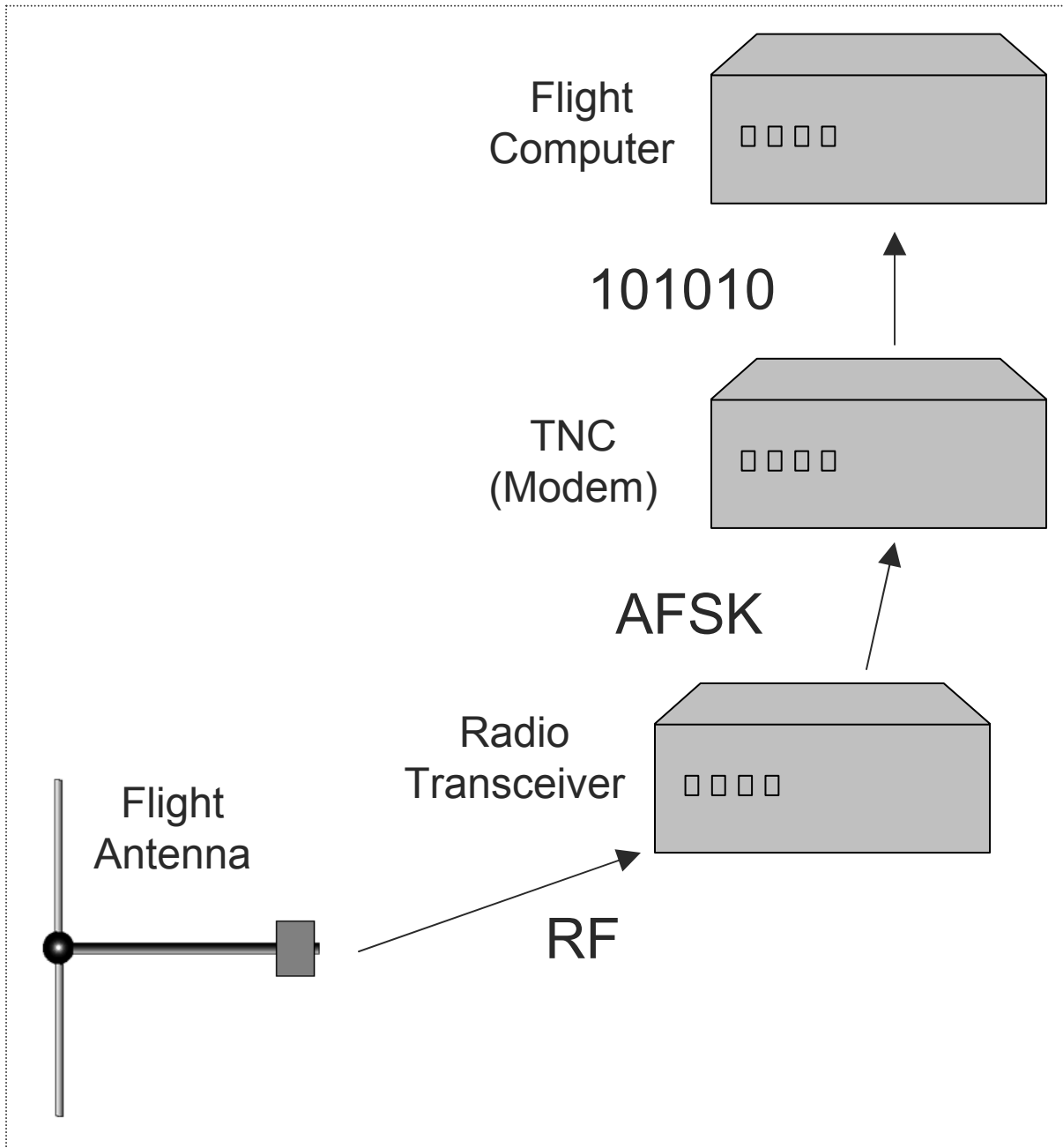
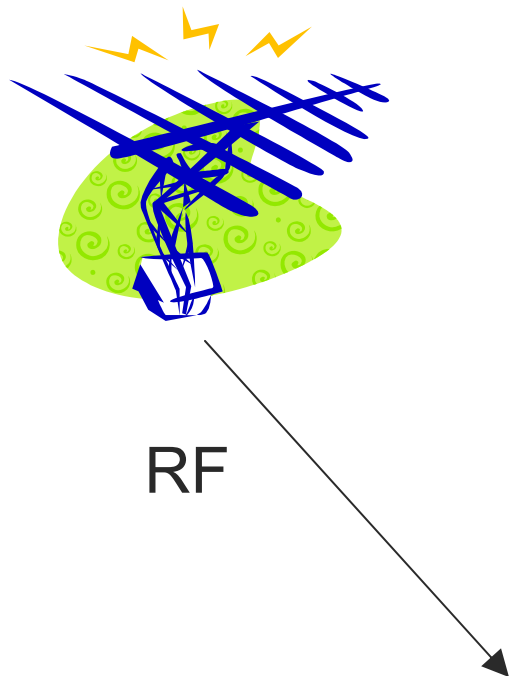


Satellite

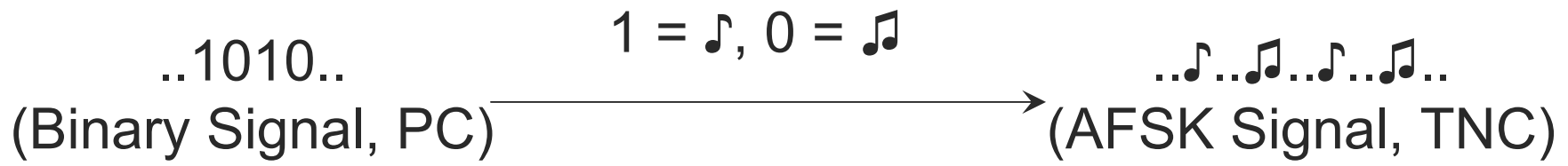


From Ground Station to Satellite

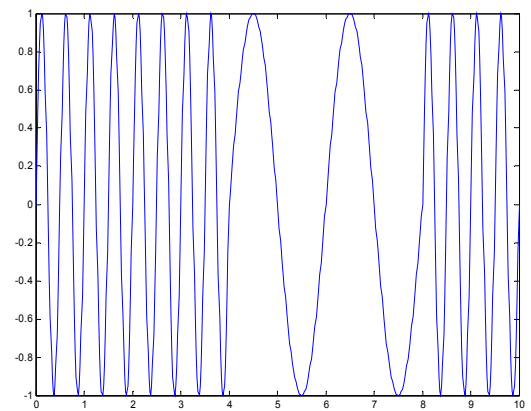
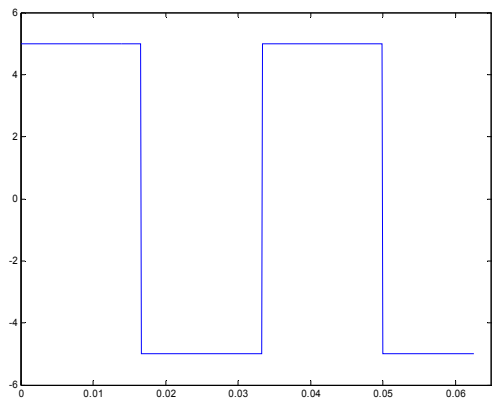




Behind the Communication: Digital Modulation

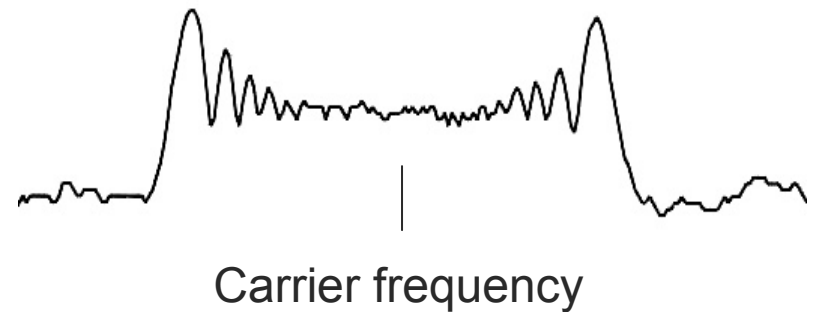
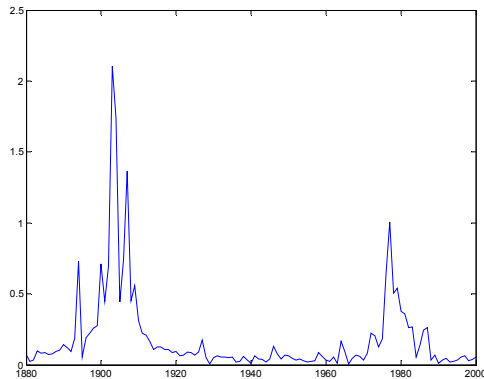
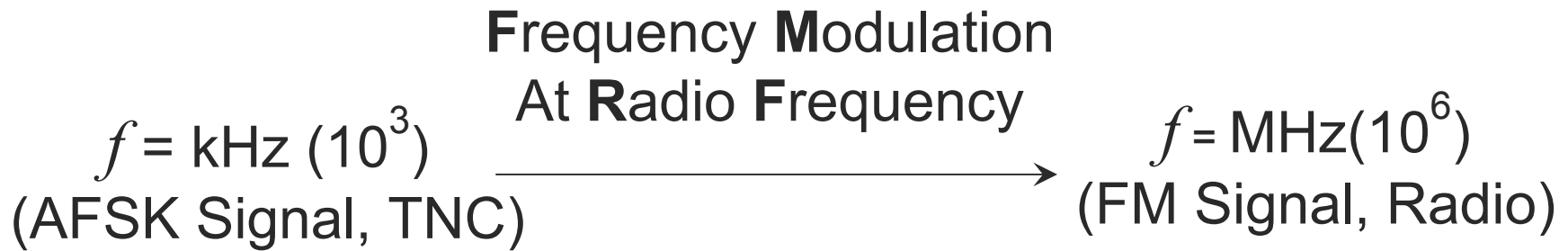


**Audio Frequency
Shift Keying**



(Time-Domain)

Behind the Communication: Analog Modulation

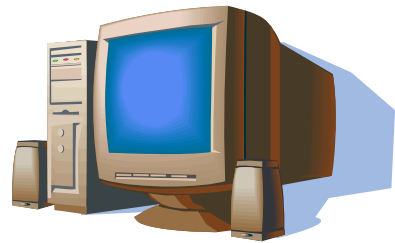


(Frequency-Domain)

[Goal]

- Make sure the communication system works.
- Measure the performance of communication system using:
 - Quality of the signal: Signal to Noise Ratio
 - Noise Level: Power of Noise Floor and path loss
 - Error Rate: Bit Error Rate
 - Transmission Rate

Basic Test



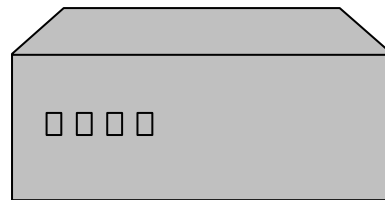
PC

101010



TNC
(Modem)

AFSK



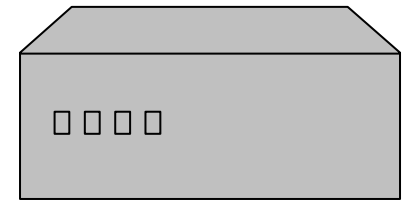
Radio
Transceiver

RF

RF

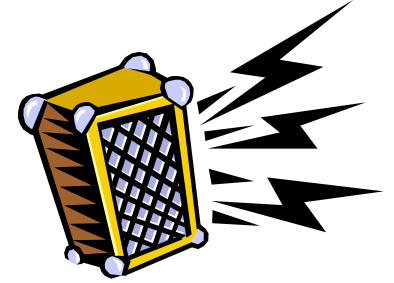
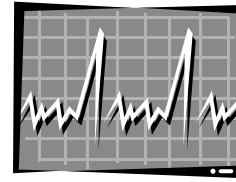


Spectrum
Analyzer

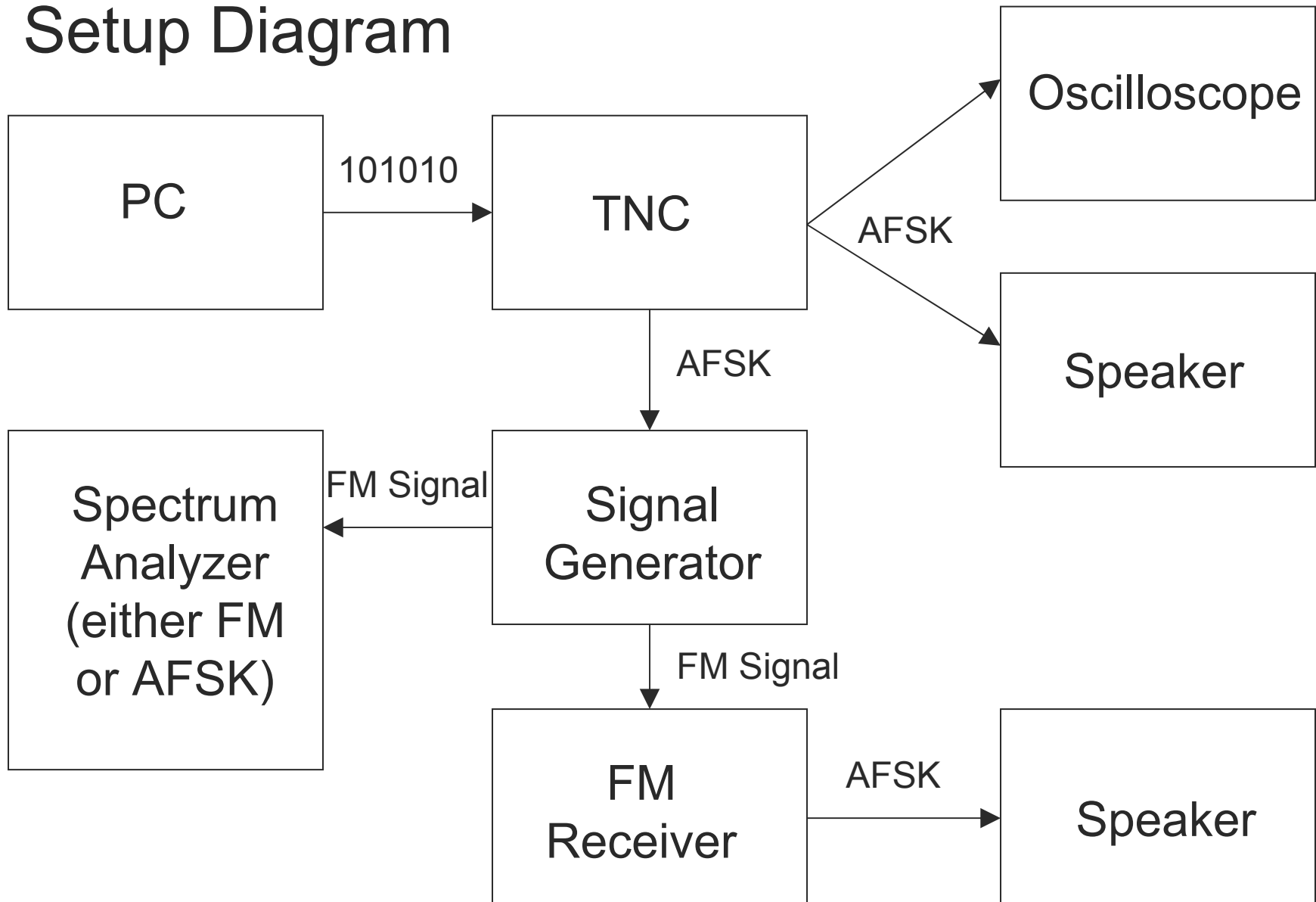


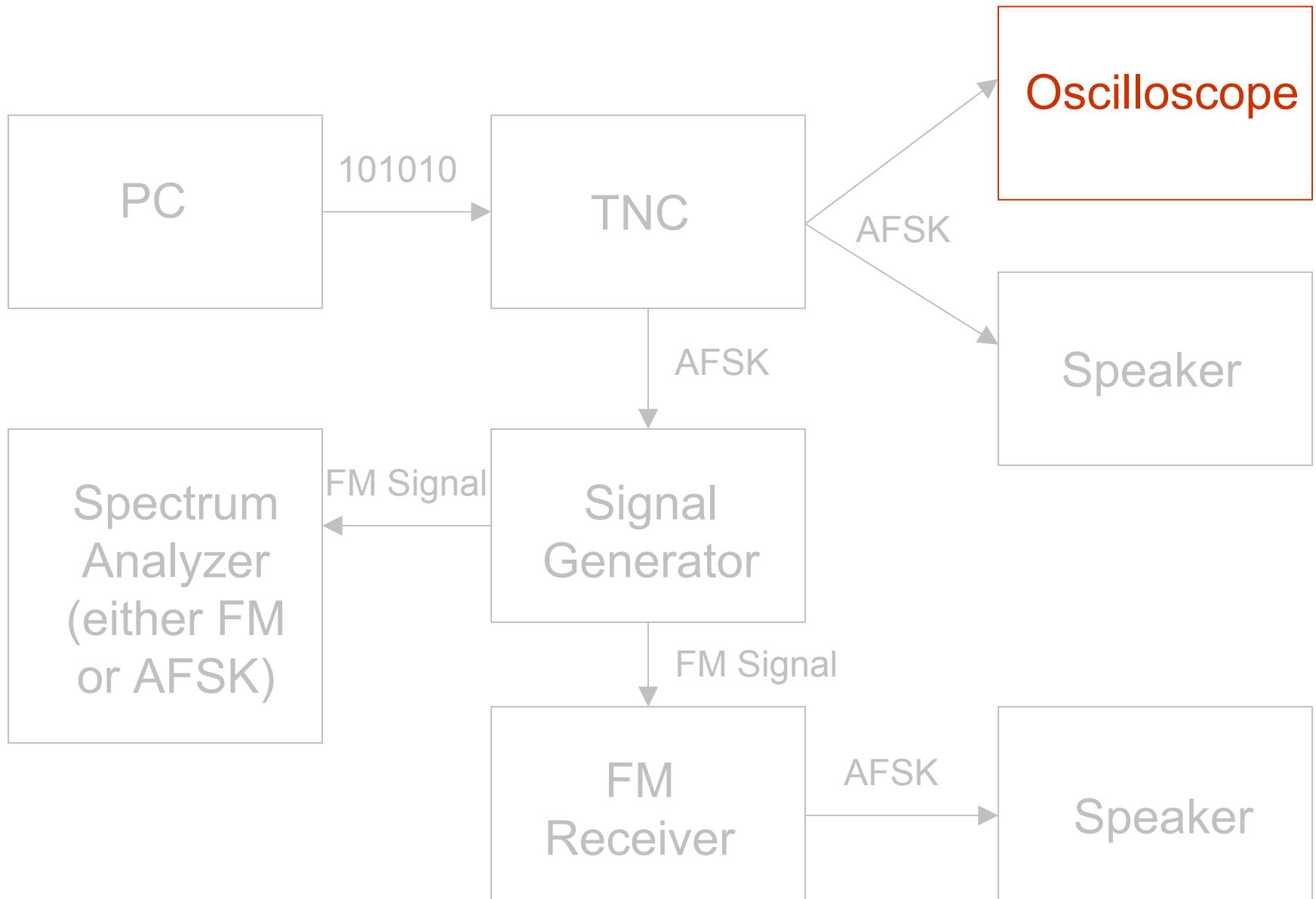
Radio
Receiver

AFSK

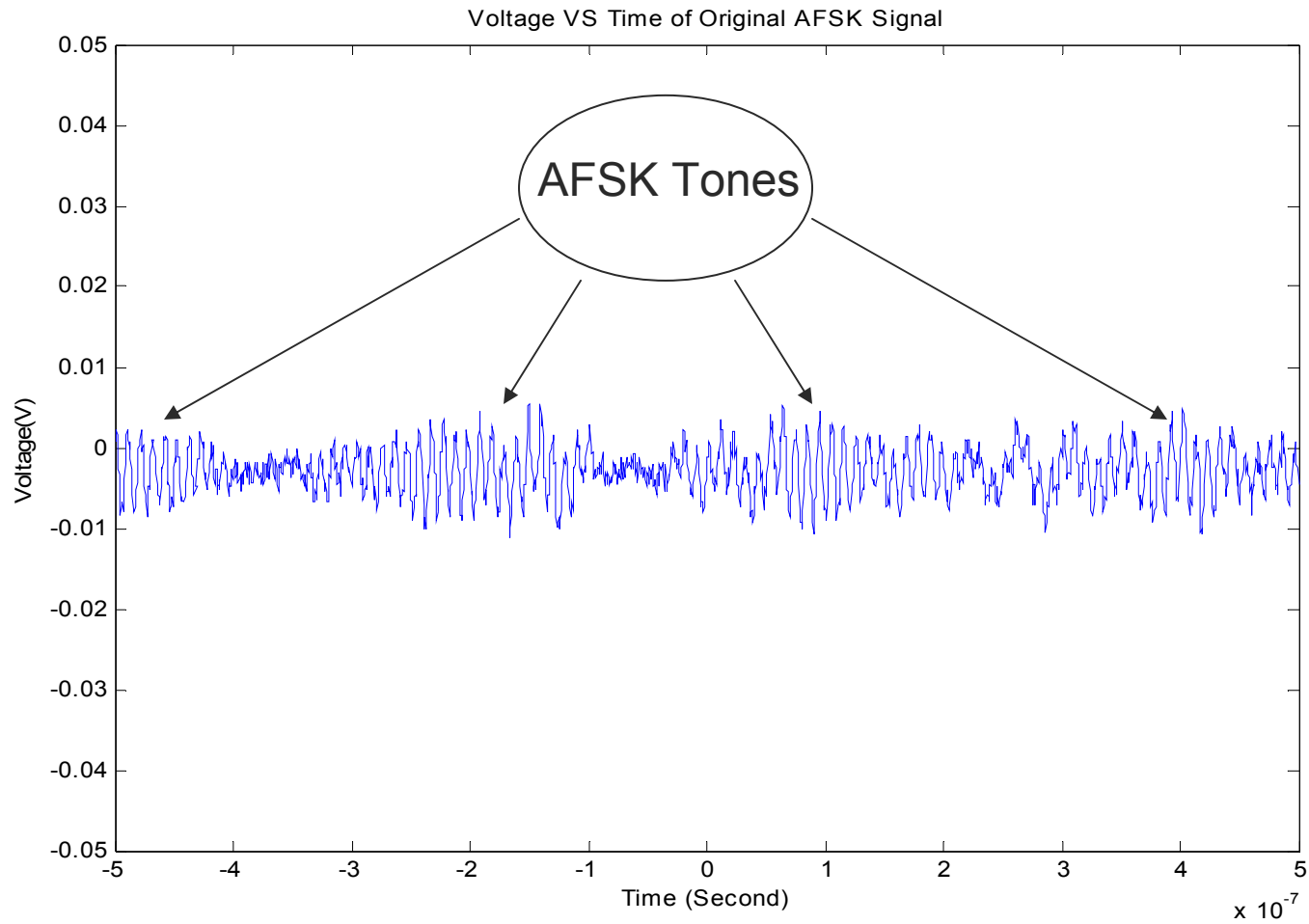


Setup Diagram

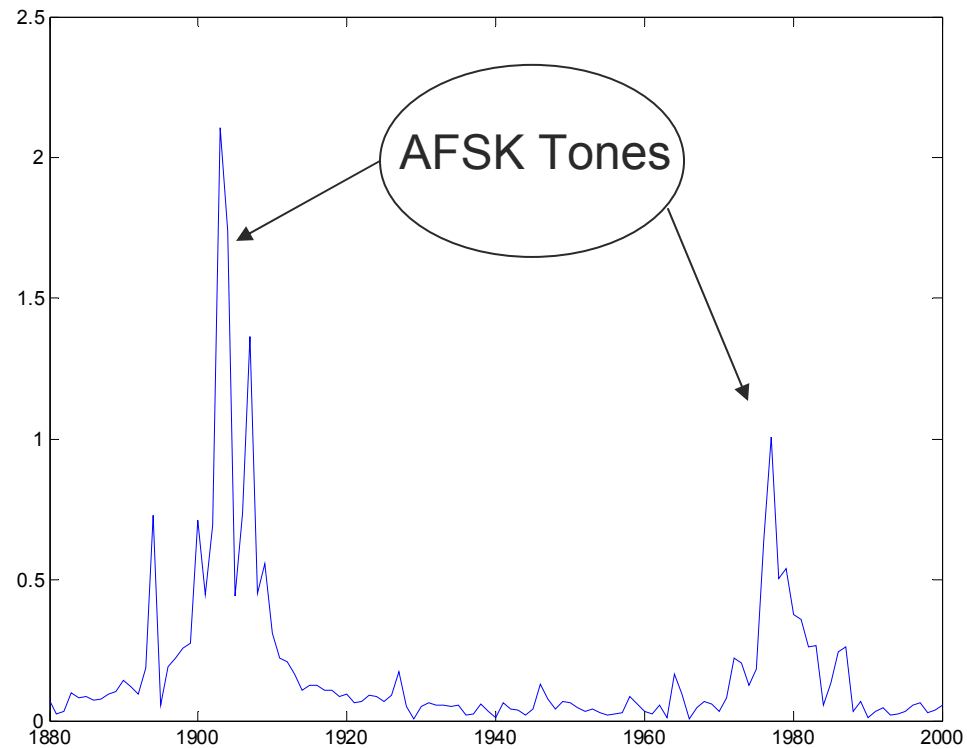


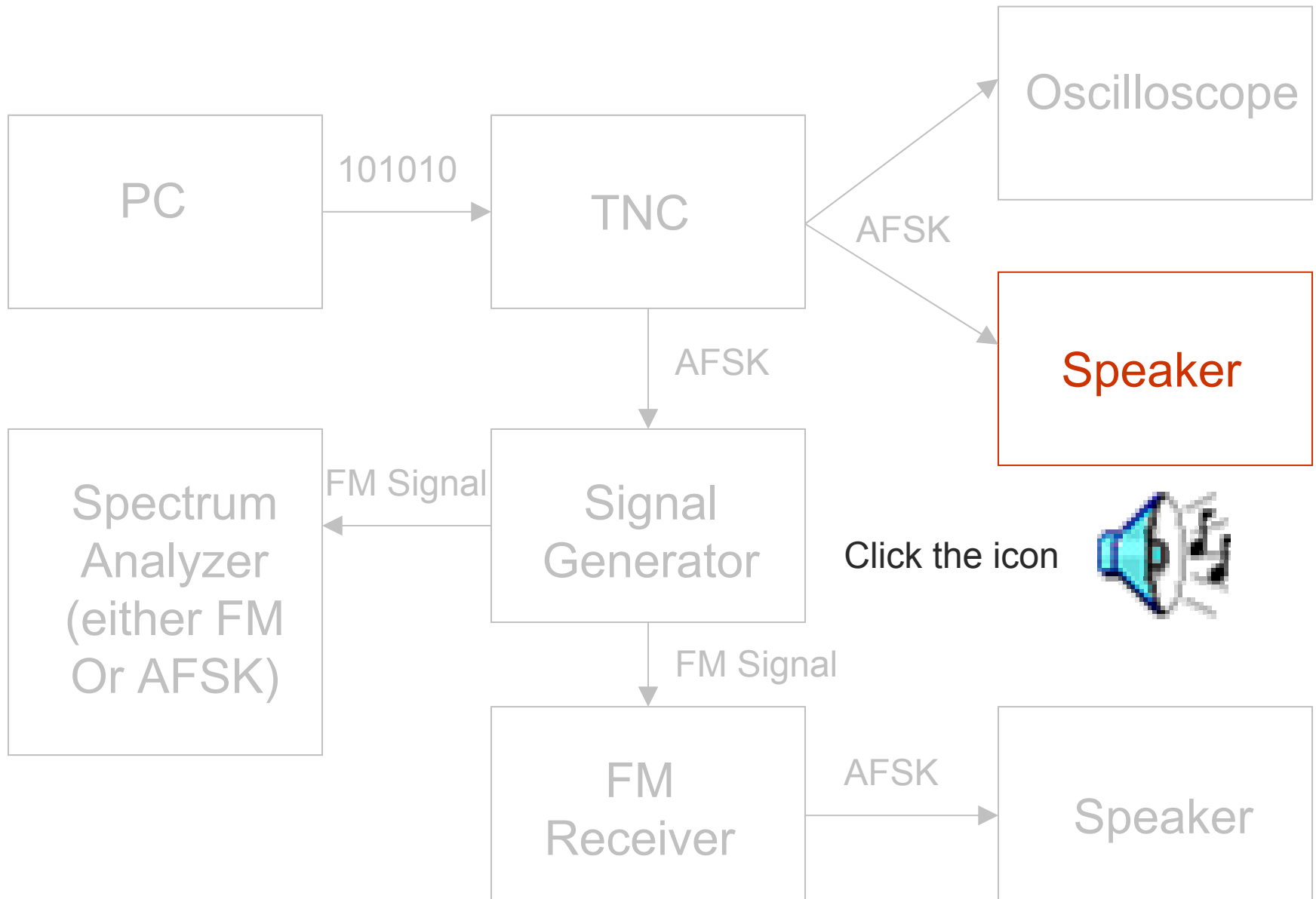


Wave form of Original AFSK Signal



Frequency Spectrum of Original AFSK Signal





Amplitude of Carrier Signal

- Let $m(t)$ be the AFSK Signal.
- After FM Modulation, the modulated signal $s(t)$ becomes:

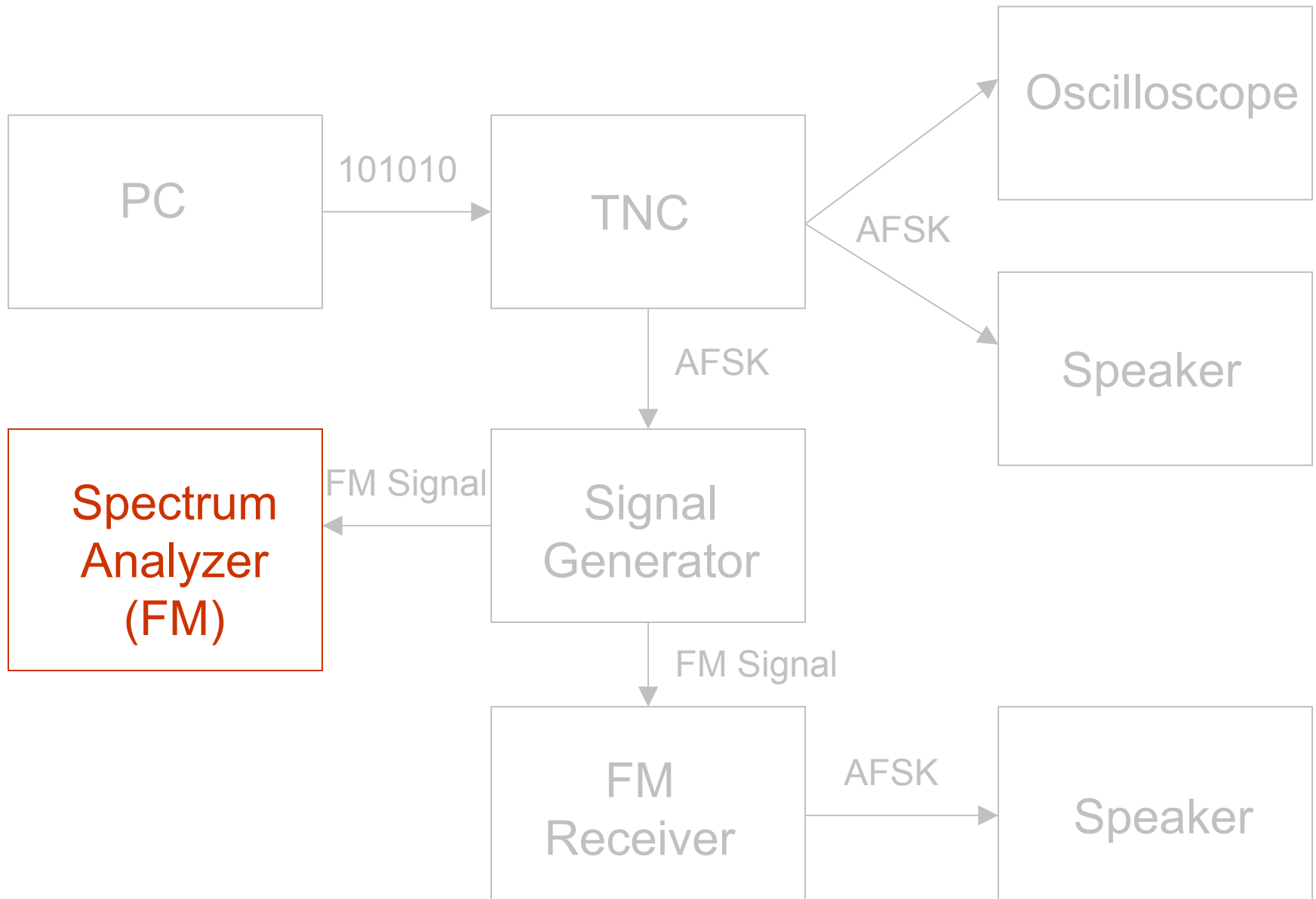
$$s(t) = A_C \cos\left(2\pi f_c t + 2\pi k_f \int_{-\infty}^t m(t') dt'\right)$$

- A_C : amplitude of the carrier signal
- f_c : frequency of the carrier signal

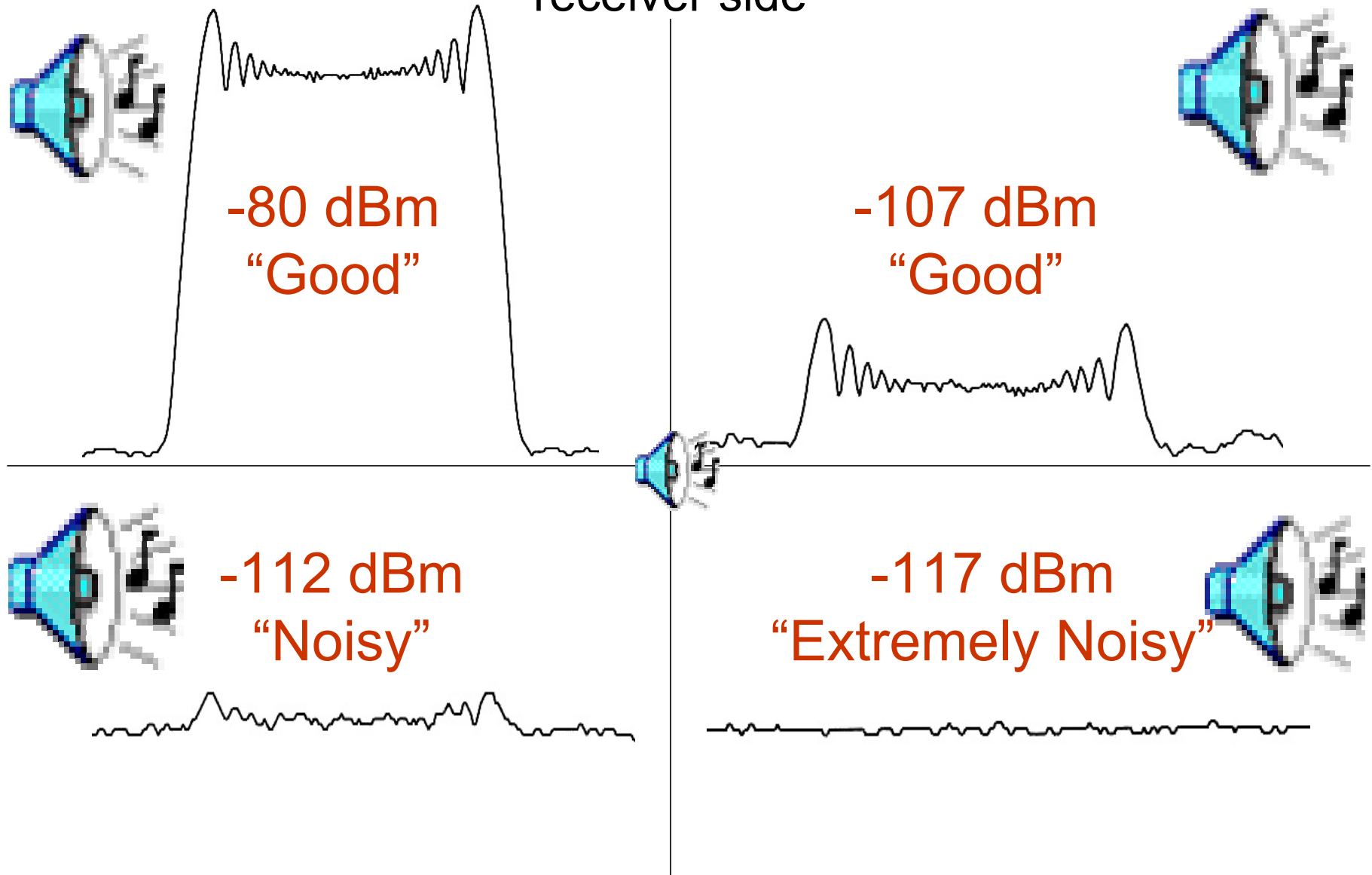
Signal to Noise Ratio (SNR)

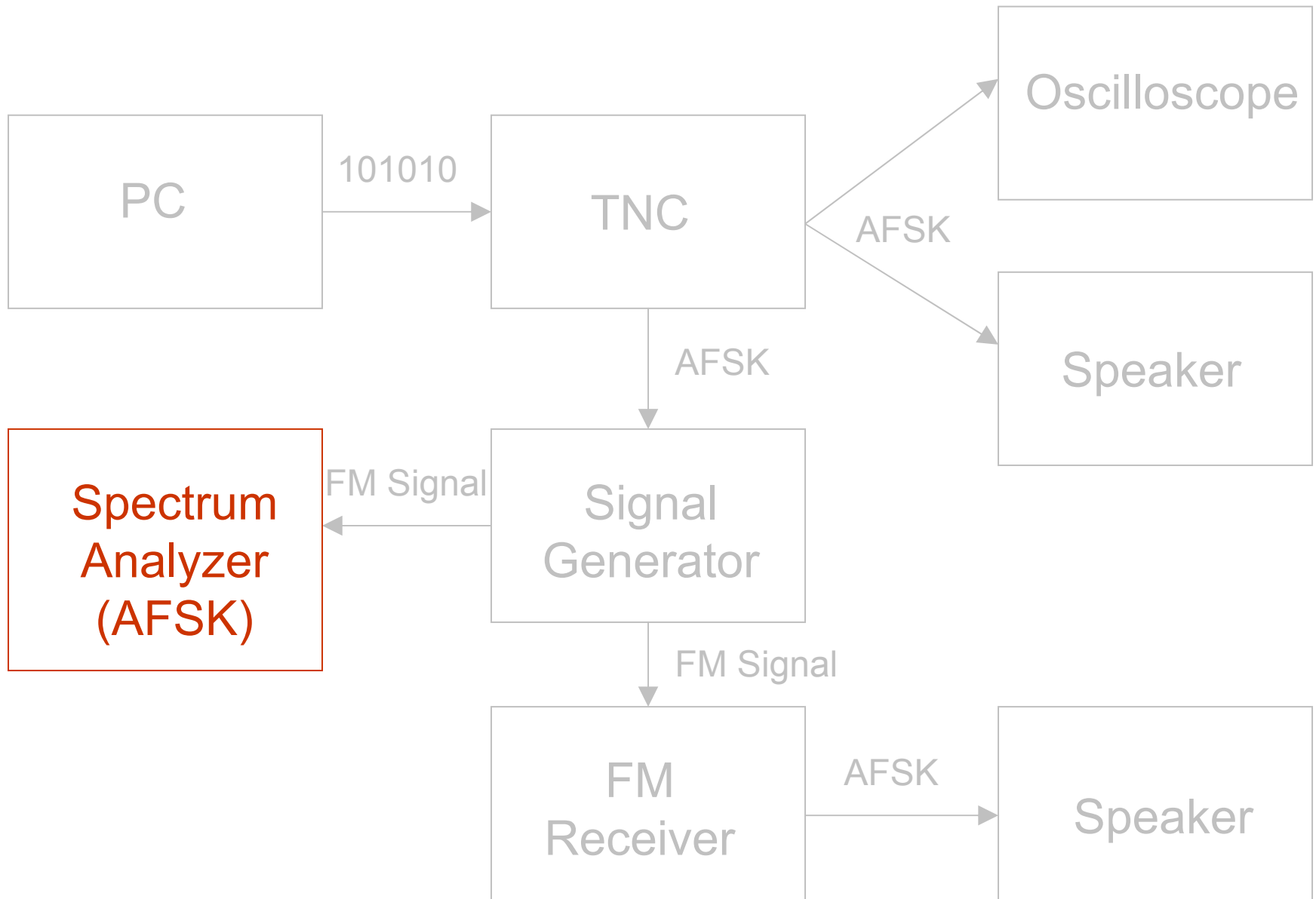
- SNR depends on A_c

$$\begin{aligned} SNR &= \frac{\text{Power of Signal}}{\text{Power of Noise}} = \frac{\int_{-\infty}^{\infty} s^2(t) dt}{\int_{-\infty}^{\infty} n^2(t) dt} \\ &= \frac{A_c^2 \int_{-\infty}^{\infty} \left(\cos\left(2\pi f_c t + 2\pi k_f \int_{-\infty}^t m(t') dt'\right) \right)^2 dt}{\int_{-\infty}^{\infty} n^2(t) dt} \end{aligned}$$



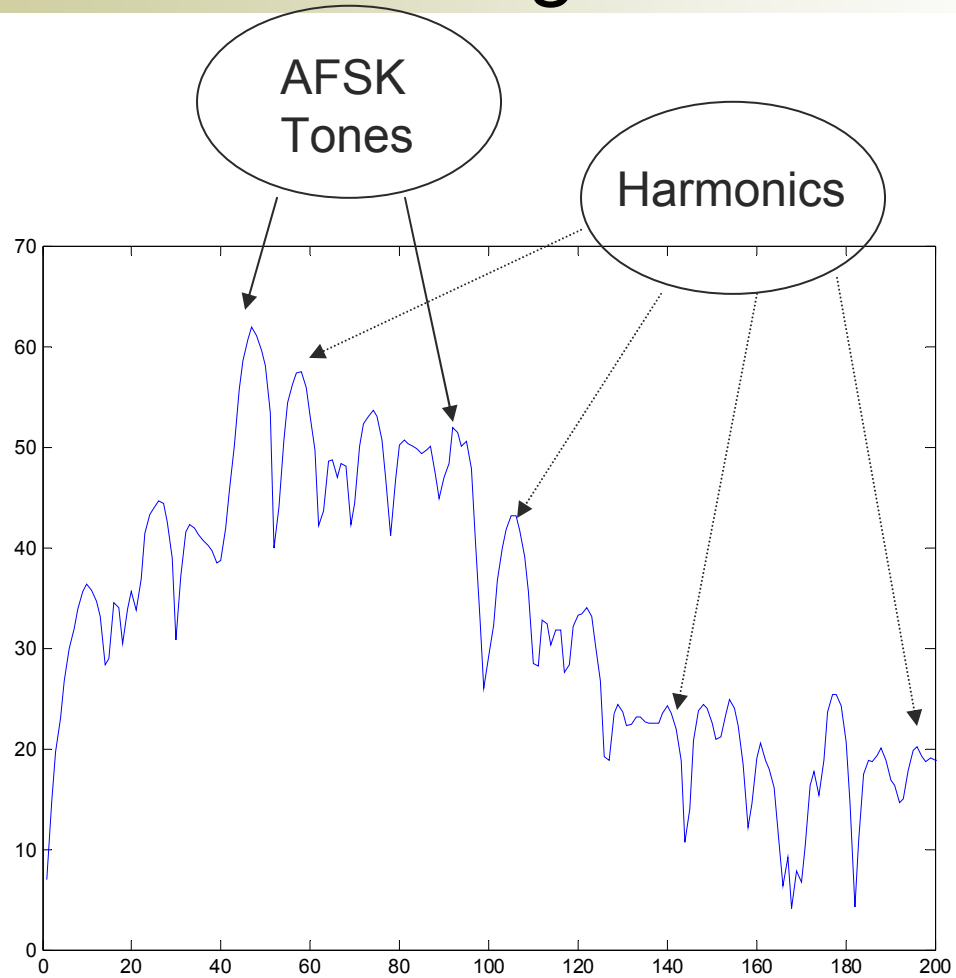
Frequency Spectrum of Modulated Signal at receiver side





Frequency Spectrum of Demodulated AFSK Signal

-80 dBm
"Good"



Summary of Important values

- $A_C > -107$ dBm: Good
- $A_C = -112$ dBm: Noisy
- $A_C < -117$ dBm: Extremely Noisy
- f_C : 435 MHz
- k_f : 15 kHz

Measurement Method

- Signal to Noise Ratio: Spectrum Analyzer
- Noise Floor: Spectrum Analyzer
- Path Loss: Spectrum Analyzer

Wireless Range Test

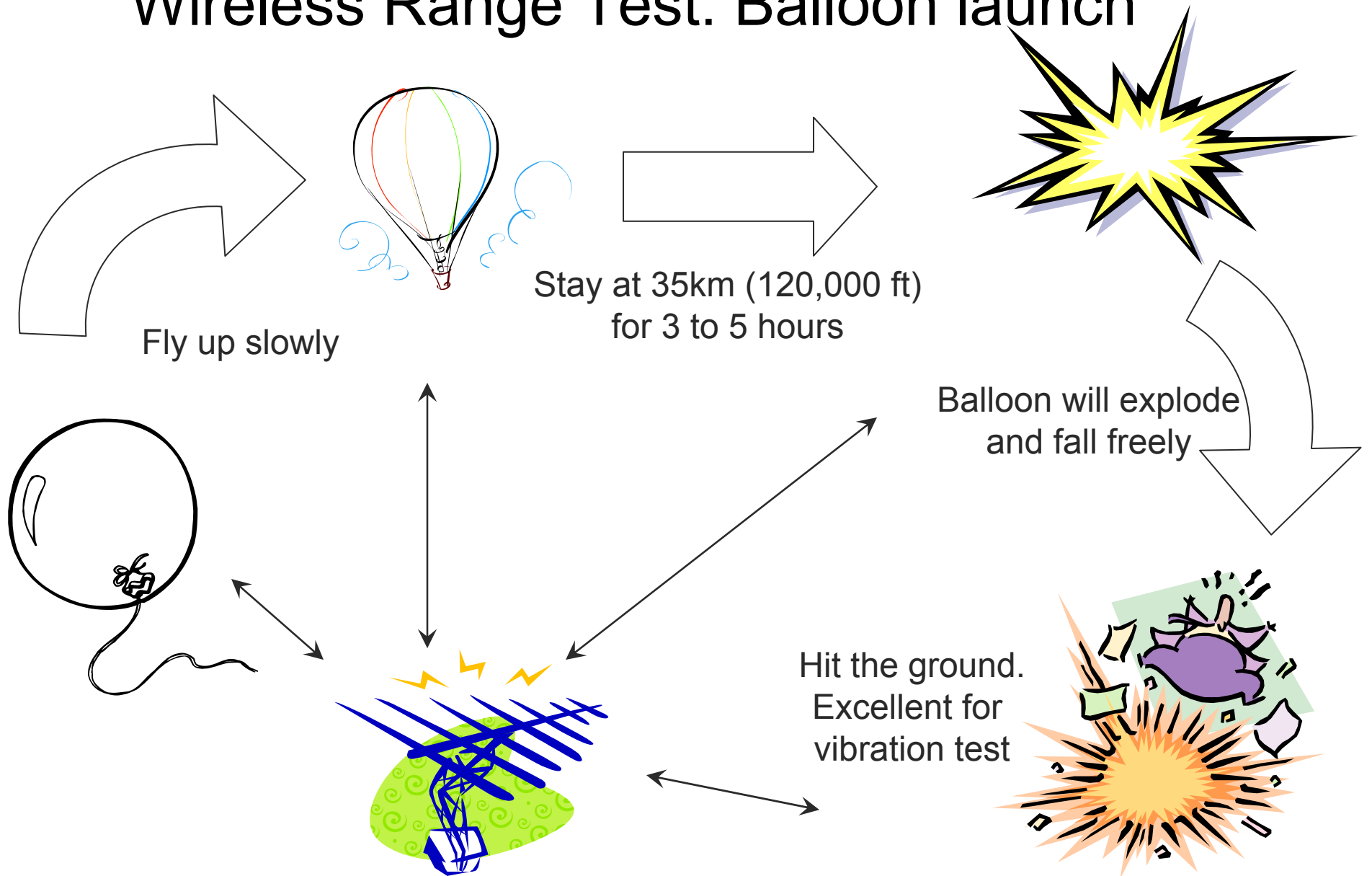
- Short Range

- Within Everitt Lab and Quad

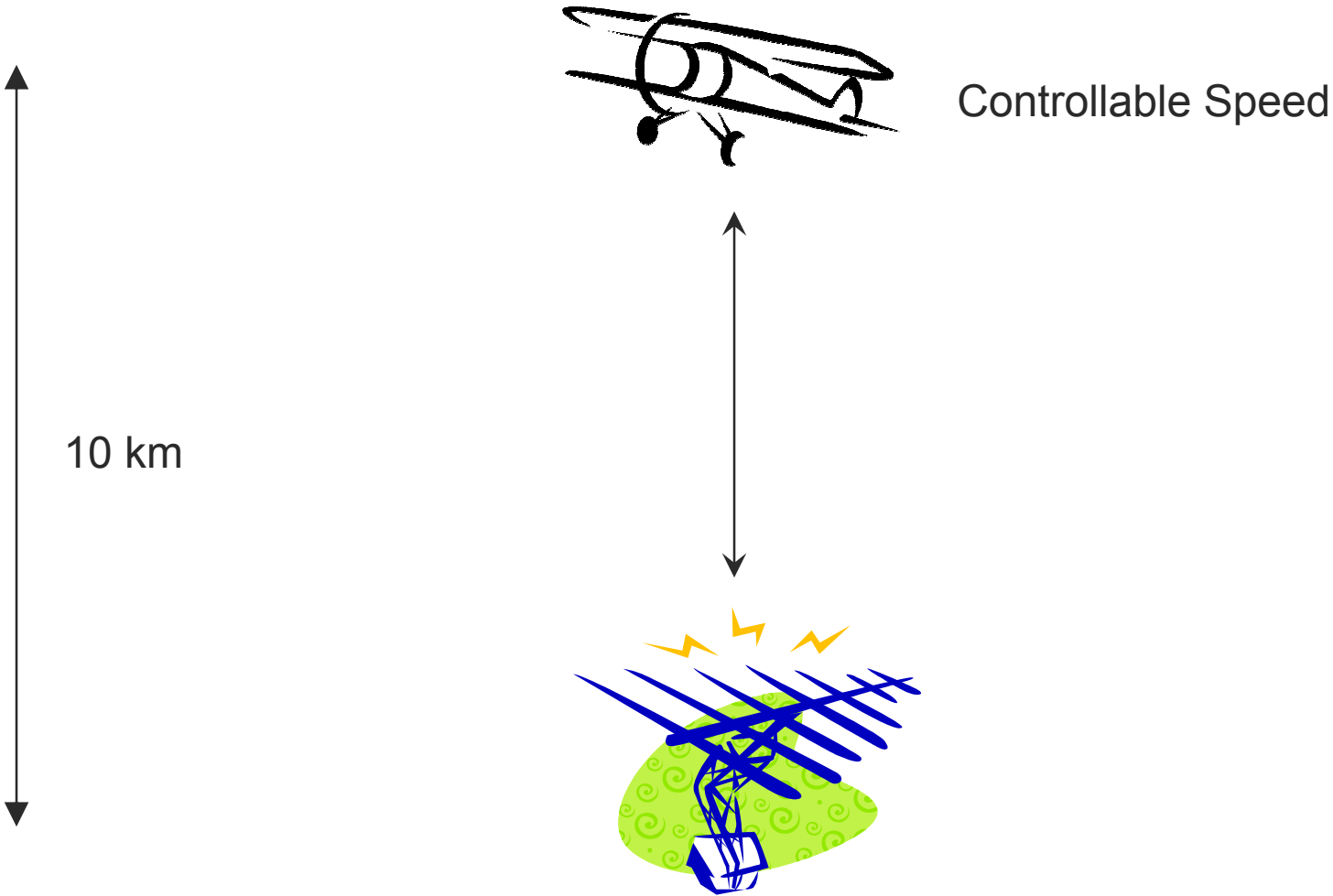
- Long Range

- Field Site: 16 km (Horizontal distance)
- Balloon Launch: 35 km (Vertical distance)
- Small Aircraft: 10 km (Vertical distance)

Wireless Range Test: Balloon launch



Wireless Range Test: Small Aircraft



Measurement details #1

- GPS will be installed on the flight unit for tracking the position.
- Temperature, pressure and height attitude will be collected.

Measurement details #2

- Signal to Noise Ratio and Path loss will be measured using Spectrum Analyzer.
- Spectrum Analyzer will installed at the receiver side. It is possible to remotely control using Agilent VEE Pro.

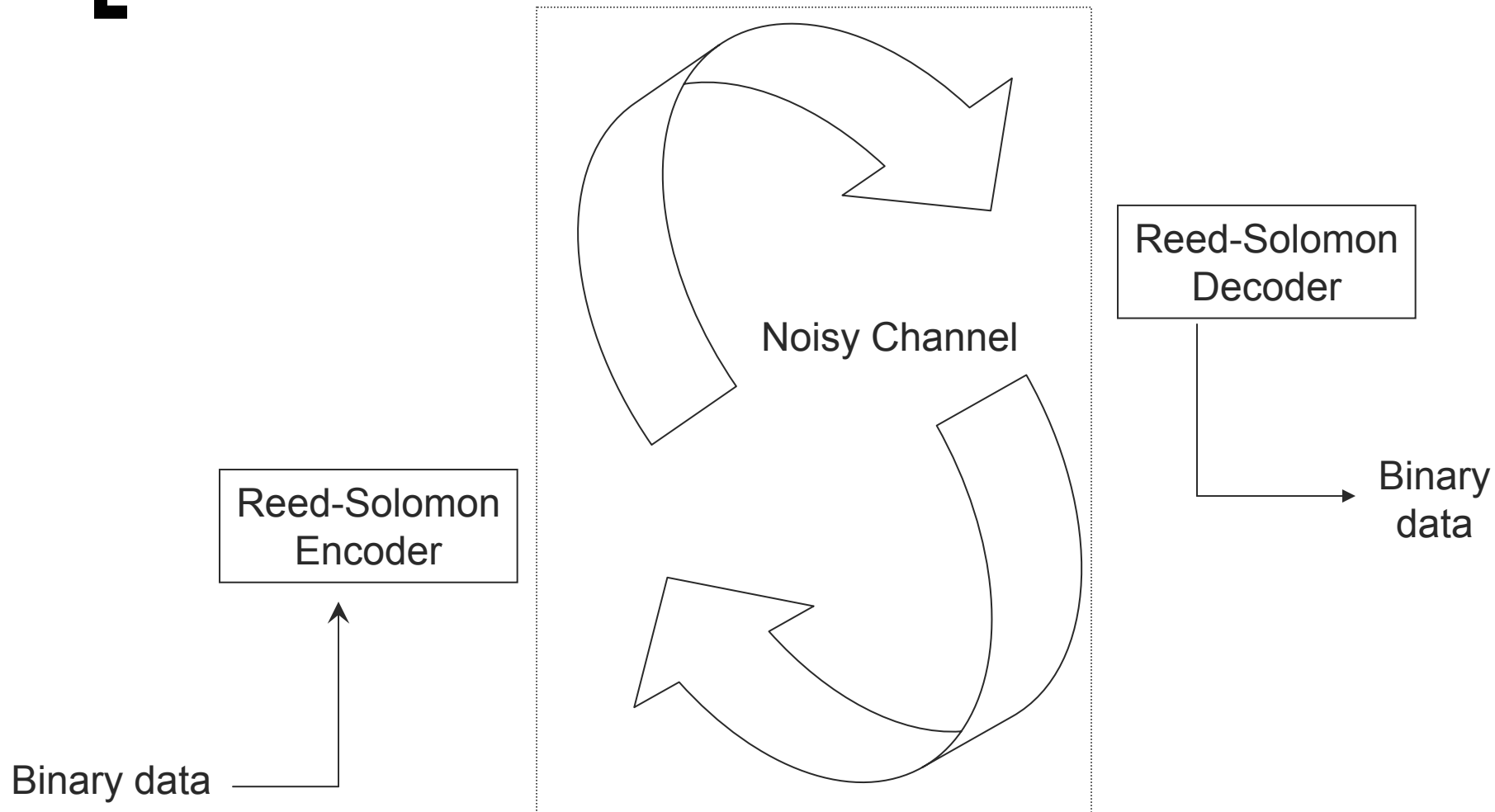
Measurement details #3

- A file will be used for analyzing the performance

$$\text{Transfer Rate} = \frac{\text{No. of bit transferred}}{\text{Total time}}$$

$$\text{Bit Error Ratio} = \frac{\text{No. of bit of error}}{\text{Total no. of bit transferred}}$$

Error Handling: Reed-Solomon Coding

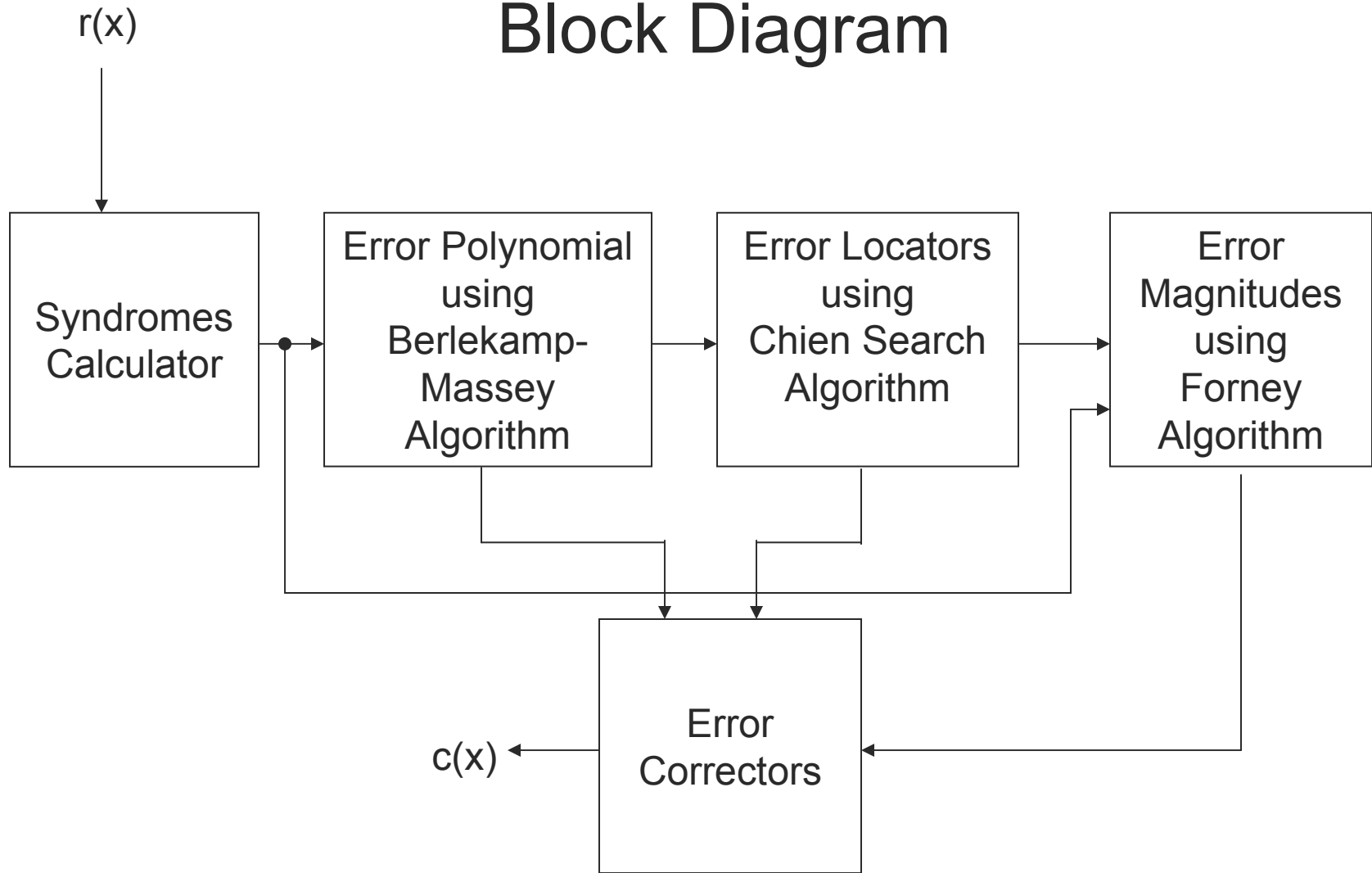


Error Handling: Reed-Solomon Coding

Let the received code word $r(x)$ is the original codeword $c(x)$ plus errors $e(x)$, i.e.,

$$r(x) = c(x) + e(x)$$

Block Diagram



~ *End* ~